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1**DE-ICED TOTAL AIR TEMPERATURE SENSOR****GENERAL FIELD**

The present invention relates to a sensor for measuring at least
5 physical parameter of a fluid flow, and in particular it relates to a de-iced
sensor of total air temperature.

A particularly advantageous application of the invention lies in the field
of aviation for measuring total air temperature admitted to aircraft engines,
and/or outside aircraft.

STATE OF THE ART

Numerous de-iced sensors of total air temperature are already known.

Conventionally, as shown in Figures 1 and 2, they comprise an air
intake 1 fitted on a streamlined body 2 having a profile of the aircraft wing
15 type.

A duct 3 is provided in the streamlined body 2 and allows the flow of
fluid having a physical parameter that is to be measured to be put into
communication with the air intake 1 via an inertial separation zone 4.

This zone 4 serves to separate elements of large mass (liquid, frost,
20 sand, ...) from the remainder of the gas by centrifuging, said elements being
evacuated from the sensor via an ejection zone 5 opposite from the air intake
1.

In order to avoid phenomena of fluid separating in the inertial
separation zone, holes 6 are made through the wall thereof on its side
25 opposite from the ejection zone 5, and in communication with the outside via
a chamber 7 which extends transversely relative to the streamlined body 2.

The pressure difference that exists between the inside and the outside
of the sensor enables the boundary layer to be sucked in through the holes 6.

The assembly comprising the air intake 1, the streamlined body 2, the
30 duct 3, the inertial separation zone 4, and the ejection zone 5 is electrically
de-iced by heater resistor elements positioned in grooves 8 formed in the
walls of the sensor.

An element 9 forming a sensing element extends inside the duct 3.

By way of example, the element 9 is a resistance thermometer that is
35 thermally insulated from the streamlined body 2.

The streamlined body 2 also referred to as a "mast" is fitted to a fixing flange 11 which is of generally plane shape (for example a disk), extending perpendicularly to the axis of the body 2 and of the duct 3.

5 A connection socket 10 is fitted to the fixing flange 11 on its side opposite from the streamlined body 2.

Provision can also be made for the streamlined body to be mounted on a moving flap to steer it onto the axis of the air flow.

10 A structure of this kind is described in particular in patent application WO 01/44821 which describes a multifunction sensor including means for measuring total air temperature mounted on a moving flap and extending therewith along a longitudinal axis that is inclined in the fluid flow.

Generally, as shown in Figure 2, the air intake 1 is of rectangular inside section and the same applies, at least over a certain fraction, to the duct 3 which is connected to the air intake 1.

15 The Applicant has recently proposed, in particular in its application WO 01/88496, a sensor structure in which the air intake is of section that is circular or elliptical, at least in part.

20 It has been shown that such a shape makes it possible to withstand icing conditions that are more severe than is possible with sensors having air intake sections that are rectangular.

SUMMARY OF THE INVENTION

25 A general object of the invention is to propose a sensor presenting behavior that is further improved under icing conditions, but without degrading measurement performance under dry conditions.

In particular, the invention relates to a sensor comprising a fluid intake fitted to a streamlined body; a duct provided in said streamlined body to enable fluid flow, said duct communicating with said fluid intake; and a sensing element disposed inside said duct.

30 The invention proposes in particular an air intake structure which enables de-icing speed to be further improved and which enables the drawbacks of air intakes that are conventionally known for this type of sensor to be mitigated.

35 With prior art sensors, difficulties are encountered in eliminating certain pieces of ice that become blocked against the inlet section of the air intake. In some cases, it can happen that the heating system on its own is not sufficient for disengaging the air intake quickly.

The invention provides a sensor for measuring a physical parameter of a fluid, in particular for measuring total air temperature, the sensor comprising:

- a fluid intake fitted to a streamlined body;
- 5 · a duct provided in said streamlined body to enable fluid flow, said duct communicating with said fluid intake; and
- a sensing element disposed inside said duct;

the sensor being characterized in that said fluid intake presents an inlet section which extends in such a manner as to define a surface that slopes
10 relative to the perpendicular to the main flow direction of the fluid at said intake, said surface defining a sliding surface for any pieces of ice that may come to bear thereagainst.

The invention also provides a solution for mitigating the drawbacks that can arise with the boundary layer suction means constituted by the holes
15 structure of the type shown in Figures 1 and 2.

Such suction means present low efficiency, which implies a large number of holes; this implies tight manufacturing tolerances in order to ensure that performance is repeatable; such means are also at the origin of the sensor having significant sensitivity to angle, which degrades
20 measurement performance when the sensor is at an angle of incidence.

The invention proposes a sensor in which the suction means do not present those drawbacks.

In particular, the invention provides a sensor for measuring physical parameter(s) of a fluid, in particular total air temperature, the sensor
25 comprising:

- a fluid intake fitted to a streamlined body;
- a duct provided in said streamlined body to enable fluid flow, said duct communicating with said fluid intake; and
- a sensing element disposed inside said duct, the fluid intake
30 presenting an inside section defined by at least one plane surface which communicates with a chamber that opens to the outside and that constitutes a boundary layer suction chamber;

the sensor being characterized in that said plane surface presents for this purpose a plurality of suction slots extending transversely relative to the
35 general flow direction of the stream in the fluid intake.

The invention also proposes improvements relating to the sensing element. Conventionally, the sensing element constituted by a resistive wire

wound between two metal tubes from which it is insulated electrically is positioned inside the sensor by means of a hollow metal support. The metal support is assembled with the metal body of the sensor by soldering. As a result, heat dissipated in the sensor body is transmitted to the sensing element by conduction.

In order to limit the effects, it is common practice to use support studs between the metal support and the sensing element.

Nevertheless, support stud structures present numerous drawbacks:

- assembly is complex;
- the support studs have poor fatigue behavior, leading to the measuring element rupturing; and
- electrical insulation is limited because metal materials are used.

The invention proposes a structure which enables those drawbacks to be mitigated by using a thermally insulating ceramic as a support mandrel providing improved mechanical strength and thermal decoupling.

Thus, the invention provides a sensor for measuring physical parameter(s) of a fluid, in particular total air temperature, the sensor comprising:

- a fluid intake fitted to a streamlined body;
- a duct provided in said streamlined body to enable fluid flow, said duct communicating with said fluid intake; and
- a sensing element disposed inside said duct;

the sensor being characterized in that the sensing element comprises a ceramic tube having a measurement resistive wire wound thereon.

BRIEF DESCRIPTION OF THE FIGURES

Other characteristics and advantages of the invention appear further from the following description which is illustrative and non-limiting and which should be read with reference to the accompanying drawings, in which:

· Figures 1 and 2, described above, are diagrammatic section and perspective views of a prior art de-iced sensor for measuring total air temperature;

· Figure 3 is a section view showing one possible embodiment of the invention;

· Figure 4 is a side view of the embodiment shown in Figure 3;

· Figures 5 and 6 are perspective views showing two examples of air intake shapes;

· Figure 7 is a diagrammatic plan view of the bottom wall of the air intake together with means for sucking in the boundary layer included therein;

· Figures 8a and 8b show the flow of the fluid firstly in suction means of the type known in the prior art, and secondly in suction means of the type shown in Figure 7;

· Figures 9a and 9b are firstly a diagram showing the angle of incidence taken by the sensor relative to the flow direction, and secondly how measurement error varies as a function of angle of incidence, firstly for a prior art sensor and secondly for a sensor of the type shown in Figures 5 to 7;

· Figures 10a and 10b are diagrammatic section views showing the flow of fluid in the thickness of the bottom wall, firstly in suction means having holes (Figure 10a), and secondly in an advantageous embodiment of the invention (Figure 10b);

· Figures 11a, 11b, and 11c show various possible shapes for the slots in a sensor of the invention;

· Figure 12 is a diagrammatic plan view similar to that of Figure 7, showing another possible embodiment;

· Figures 13a and 13b show two possible embodiments for the sliding surface defined by the inlet section of the air intake;

· Figure 14 is a diagrammatic cross-section view of the air intake;

· Figure 15 is a diagrammatic perspective view of a sensor constituting a possible embodiment of the invention, this perspective view showing one possible shape for the ejection section of the sensor; and

· Figure 16 is a diagrammatic side view of the sensing element.

DESCRIPTION OF EMBODIMENTS OF THE INVENTION

General structure

The sensors shown in Figure 3 et seq. have an air intake 1 supported by a streamlined body or "mast" 2 and opening out into a duct 3 which extends inside said mast 2, the duct 3 receiving a sensing element 9.

The streamlined body or "mast" 2 extends from a fixing flange 11.

The fixing flange presents, in particular, a bearing surface 11a for bearing against the part on which the sensor is to be fixed (e.g. a portion of fuselage), this bearing surface defining a fixing plane for the sensor.

Inclination and shape of the inlet section

Conventionally, the section supporting the air intake of the sensor is perpendicular to the flow.

As shown in Figure 13a, it is proposed that the inlet section of the air intake should be inclined relative to the section orthogonal to the flow by an angle θ .

The inclination is from the bottom edge of the intake going downstream (backwards) so that the inlet section of the air intake constitutes a sliding surface for pieces of ice that become blocked against said air intake.

Such pieces of ice are caused to slide along said surface and to disengage it, said surface being heated by the de-icing system.

With such a structure having an inclined entry section the speed of de-icing is improved by up to 50% (when the angle θ lies in the range 5° to 45°). The angle θ is preferably equal to 10° , in particular.

In another variant, and as shown in Figure 13b, the entry section which forms a sliding surface for ice can be rounded (concave or convex).

In order to optimize both the speed of de-icing and the efficiency of the suction slots, the shape of the air intake is preferably of the type shown in Figure 6, and also as shown in Figure 14.

As can be seen in these figures, this shape comprises two plane sides that are mutually parallel and of different lengths in right section of the air intake. These sides are connected together by two sides of rounded shape.

The side which defines the bottom plane 1b and which carries the suction slots 12 is, for example, of a length lying in the range 15 millimeters (mm) to 25 mm (double-headed arrow referenced L in Figure 14), while the second side is of a length lying in the range 5 mm to 15 mm (double-headed arrow referenced ℓ in Figure 14).

The other two sides which are rounded serve to reduce the inside surface area that is exposed to icing. They have a radius of curvature (R) lying in the range 5 mm to 15 mm.

It can also be observed that all of the angles between the various sides of the right section of the air intake are greater than 90° , thus making it possible to limit dead zones and to have a uniform distribution of fluids.

As shown more particularly in Figures 5 and 6, the air intake 1 is of a shape that is rounded, at least in part.

In the example shown in Figure 5, the air intake 1 is defined by a top portion 1a of inside shape that is substantially cylindrical, and by a bottom

wall 1b that is plane in shape, extending between the edge of the opening to the air intake 1 and the inertial zone 4, where the terms "top" and "bottom" should be understood herein as being relative to the main axis of the body 2 with the fixing flange 11 being taken as being at the bottom end and the air intake at the top end. The duct 3 also presents a rounded shape, at least in part.

The use of rounded shapes for the air intake 1 and/or for the duct 3 has the advantage of enabling the inside surface area of the sensor on which ice might be deposited to be reduced, and of eliminating zones in which the flow section of the air or fluid to be measured changes, thereby eliminating the dead zones that are generated in corners.

For given de-icing power, these shapes enable icing conditions to be withstood that are more severe than is possible with conventional sensors having an air intake of rectangular-shaped section; they also make it possible to comply with the latest developments in aviation standards. In particular, for identical icing conditions, the de-icing power required is reduced by 10% to 20% compared with prior art sensors.

Nevertheless, it should be observed that air intakes of the type shown in Figure 6 are more particularly preferred: in particular the plane top surface 1c makes it possible to ensure an internal compression ratio that is sufficient to ensure good efficiency for the suction system, regardless of the external flow rate.

Suction of the boundary layer

In the example shown in Figure 5, as in Figures 6 and 7, the plane bottom surface 1b of the air intake 1 does not have the suction holes that are usually provided for systems for sucking in boundary layers, but instead has slots 12 which extend perpendicularly to the flow direction of the stream through the air intake.

These slots 12 serve to limit development of a thermal boundary layer inside the sensor and also inside the duct 3 which contains the sensing element, because:

- of an increase in suction efficiency due to a larger effective section area; and
- of an absence of flow lines being deflected, said slots making it possible for the flow to present a certain amount of width (flow 2D in Figure 8b) unlike the suction system based on holes (Figure 8a).

A particularly advantageous application of this improvement lies with a sensor presenting an angle of incidence relative to the flow direction of the stream that is large (angle of incidence greater than 5°). The angular sensitivity when the heating system is active is then reduced by an amount
5 lying in the range 20% to 50%.

This is shown in Figures 9a and 9b, where Figure 9a defines the angle of incidence taken by the sensor relative to the flow direction (cf. double-headed arrow which represents pivoting about the axis of the sensor), while Figure 9b is a graph having curves plotted thereon to show how error in
10 measuring total air temperature varies as a function of angle of the incidence between the flow direction and the axis of the sensor, firstly for a prior art sensor having suction holes (continuous line curves), and secondly for a sensor provided with suction slots (dashed line curves).

It can be seen that with suction slots, angular behavior is greatly
15 improved, the sensor being particularly insensitive to the effects of angle, the separation angle of the sensor being in any event pushed back a large amount.

It should also be observed that the use of slots makes it possible to eliminate the cold points that are present between the suction holes used in
20 the prior art and that are liable to encourage retention of ice.

In addition, they are less likely to become obstructed in the event of a foreign body penetrating into the sensor (sand, volcanic dust, ...).

It should also be observed that the slots may advantageously be inclined through the thickness of the bottom wall 1b. This is shown in
25 Figures 10a and 10b, which show respectively the flow of a fluid in a slot that is not inclined (Figure 10a) and in a slot that is inclined (Figure 10b).

It will be understood from these diagrams that an angle of inclination ϕ relative to the perpendicular to the wall 1b enables optimized flow efficiency to be achieved in particular by reducing zones of reverse flow inside the slots
30 and increasing the value of the feed pressure.

By way of example, the angle ϕ may lie in the range 10° to 60° . For example it is about 45° .

Furthermore, as shown in Figures 11a, 11b, and 11c, various different shapes can be envisaged for these slots. In particular, the slots may be of
35 rectangular section (Figure 11a); they may be of a section having rounded ends (Figure 11b); they may be in the form of chevrons (Figure 11c); etc.

Figure 12 shows a wall having three chevron-shaped slots 12. The tips of the chevrons 12 point towards the ejection zone 5 of the sensor.

Furthermore, these suction slots 12 open out into a chamber 7 which is defined firstly by the wall 1b and secondly by the wall 13. This inclined wall 13 enables the suction capacity of the slots 12 to be increased by optimizing the pressure difference between the inside of the sensor and the openings defined by the shape of the chamber 7, with this being achieved by generating a marginal vortex on the edges of the wall 13. This reduces the error associated with the heating system at low flow rate.

Furthermore, the ejection section of the zone 5 (Figure 15) is defined so as to allow inertial separation of particles that penetrate into the sensor and to give the sensor good ability to ingest ice crystals and frost.

Sensing element

As shown in Figure 15, the sensing element 9 is constituted by two main portions of cylindrical shape (a ceramic tube 21 and a support mandrel 22). These shapes enable energy to be exchanged in optimum manner with the fluid under measurement and consequently serve to reduce the error caused by heat transfer by conduction between the sensor and the sensing element. This effect is amplified by using a thermally insulating ceramic for the support mandrel.

Specifically, the fact of using ceramics for making the support of the measurement element provides improved mechanical strength and thermal decoupling.

It should be observed that such a structure presents improved resistance to fatigue.

Furthermore, such a structure makes it possible to guarantee electrical insulation that is better than that provided by technology relying on a metal support. In particular, the ceramic provides protection against lightning, which is not possible with metal supports.

In addition, the ceramic support provides effective thermal decoupling, even at low speed.

The sensing element 21 is advantageously constituted by a ceramic tube having a resistive wire wound thereon.

The above-described improvements are advantageously applied to a sensor which includes a fixing flange presenting a bearing surface defining a fixing plane for the sensor and in which the streamlined body is inclined

relative to the fixing plane and presents a longitudinal axis that extends other than perpendicularly relative to said plane.

More precisely, the sensor then extends so that the opening of the intake 1 lies in the fluid flow with the mast also being in the fluid, its main midplane being parallel to the flow direction thereof, the axis of said mast (i.e. the axis of the duct 3 and of the sensing element 9 referenced A in Figure 3) being inclined relative to the fluid flow and being disposed at a certain angle relative to a direction that extends in said midplane perpendicularly to the fluid flow and to the fixing plane defined by the bearing surface 11a.

This inclination is "backwards", the head of the sensor, i.e. the air intake 1, being further back than the portion of the mast 2 that is furthest away therefrom relative to the fluid flow direction.

The angle between the axis A of the mast 2 and the above-mentioned perpendicular direction (angle α in Figure 3) is about 5° to 15°.